
Singular Perturbation Methods Formation of Shock Waves

Plane waves (small amplitude) in the absence of dissipation propagate without change in shape:

$$\frac{\partial^2 u}{\partial t^2} = a_1^2 \frac{\partial^2 u}{\partial x^2} \quad (0.1)$$

$$u(x, t) = F(x - a_1 t) + G(x + a_1 t) \quad (0.2)$$

u = particle velocity
 a_1 = local speed of sound

For finite amplitude plane waves, with dissipation

$$c = a_1 \pm \frac{\gamma + 1}{2} u \quad (0.3)$$

$$c = a_1 \left\{ 1 \pm \frac{\gamma + 1}{\gamma - 1} \left[\left(\frac{\rho}{\rho_1} \right)^{\frac{\gamma - 1}{2}} - 1 \right] \right\} \quad (0.4)$$

c = wave speed

Regions of higher condensation, $\frac{\rho}{\rho_1} > 1$, overtake those of lower condensation.

- Produces steepening effect
- Non-linear convective terms \Leftrightarrow diffusive terms
- Wave becomes "stationary"

Two time scales:

- (A) Viscous diffusive terms balance steep gradients generated by piston initially
- (B) Non-linear convective terms balance viscous diffusive terms

Model:

Continuum flow formulation (Navier-Stokes)

ϵ = piston mach number, $\epsilon \ll 1$

Boundary conditions for large time: matching principle of inner and outer expansions

Gas is viscous and heat conducting

Non-dimensionalization:

$$\mu^* = \epsilon \sqrt{RT_0} \mu \quad (0.5)$$

$$\rho^* = \rho_0^* (1 + \epsilon \rho) \quad (0.6)$$

$$p^* = p_0^* (1 + \epsilon p) \quad (0.7)$$

$$T^* = T_0^* (1 + \epsilon T) \quad (0.8)$$

$$\mu^* = \mu_0^* (1 + \epsilon \mu) \quad (0.9)$$

$$x^* = \left[\frac{\mu_0^*}{\rho_0^* \sqrt{RT_0^*}} \right] x \quad (0.10)$$

$$t^* = \left[\frac{\mu_0^*}{\rho_0^* RT_0^*} \right] t \quad (0.11)$$

()* — dimensional variable

R — gas constant

(\cdot)₀ — undisturbed value

Navier-Stokes Equations

$$\rho_t + \mu_x + \epsilon(\rho\mu)_x = 0 \quad (0.12)$$

$$\mu_t + p_x - \mu_{xx} + \epsilon \left[\rho\mu_t + \mu\mu_x - (\mu\mu_x)_x \right] + \epsilon^2 \rho\mu\mu_x = 0 \quad (0.13)$$

$$T_t + (\gamma-1)\mu_x - \frac{\gamma}{\nabla} T_{xx} + \epsilon \left[\rho T_t + \mu T_x + (\gamma-1)p\mu_x - (\gamma-1)\mu_x^2 - \frac{\gamma}{\nabla} (\mu T_x)_x \right] + \epsilon^2 \left[\rho\mu T_x + (\gamma-1)\mu\mu_x^2 \right] = 0 \quad (0.14)$$

$$p = \rho + T + \epsilon \rho T \quad (0.15)$$

ρ = specific heat ratio

∇ = Prandtl number

Initial conditions

$$\mu = \rho = p = T = 0; x > 0, t = 0$$

Boundary conditions (at piston)

$$\mu = 1, T_x = 0; x = \epsilon t, t > 0$$

At infinity, damping conditions

$$\mu, \rho, T \rightarrow 0, t > 0, x \rightarrow \infty$$

Expansion:

$$\mu^o = \mu_0^o + \epsilon \mu_1^o + \epsilon^2 \mu_2^o + \dots \quad (0.16)$$

Linearized solutions, $\epsilon \rightarrow 0$ (small times) ("outer region")

$$\rho_t^o + \mu_x^o = 0 \quad (0.17)$$

$$\mu_t^o + p_x^o - \mu_{xx}^o = 0 \quad (0.18)$$

$$T_t^o + (\gamma-1)\mu_x^o - \frac{\gamma}{\delta} T_{xx}^o = 0 \quad (0.19)$$

$$p^o = \rho^o + T^o \quad (0.20)$$

Initial and boundary conditions

$$\mu^o = \rho^o = T^o = 0, x > 0, t = 0$$

$$\mu^o = 1, T_x^o = 0, x = 0, t > 0$$

$$\mu^o, \rho^o, T^o \rightarrow 0; t > 0, x \rightarrow \infty$$

Using Laplace transforms:

$$\bar{\mu}^o(x, s) = \int_0^\infty e^{-st} \mu^o(x, t) dt \quad (0.21)$$

$$\mu^o(x, t) \sim \frac{1}{2} \operatorname{erfc} \left[\frac{(x - \sqrt{\gamma}t)}{\sqrt{2\beta t}} \right] + o(t^{-\frac{1}{2}}) \quad (0.22)$$

$$\rho^o(x, t) \sim \frac{1}{\sqrt{\gamma}} \mu^o(x, t) + o(t^{-\frac{1}{2}}) \quad (0.23)$$

$$T^o(x, t) \sim \frac{(\gamma - 1)}{\sqrt{\gamma}} \mu^o(x, t) + o(t^{-\frac{1}{2}}) \quad (0.24)$$

$$\beta \equiv 1 + \frac{\gamma - 1}{\nabla} \quad (0.25)$$

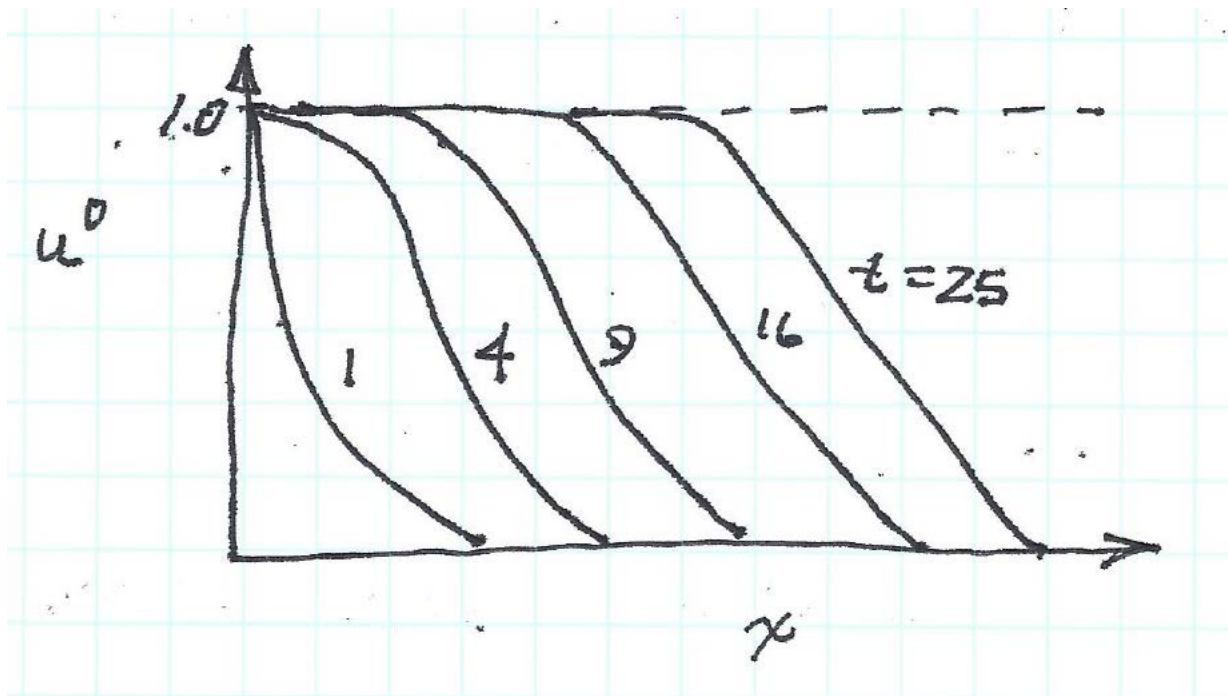
Transformed (x, t)

$$X = (x - \sqrt{\gamma}t) / \sqrt{2\beta} \quad (0.26)$$

$$\mu^o(x, t) \sim \frac{1}{2} \operatorname{erfc} \left(\frac{X}{\sqrt{t}} \right) + o(t^{-\frac{1}{2}}) \quad (0.27)$$

$$\rho^o(x, t) \sim \frac{1}{\sqrt{\gamma}} \mu^o(x, t) + o(t^{-\frac{1}{2}}) \quad (0.28)$$

$$T^o(x, t) \sim \frac{\gamma - 1}{\sqrt{\gamma}} \mu^o(x, t) + o(t^{-\frac{1}{2}}) \quad (0.29)$$



SOLUTION AT LARGE TIMES ("INNER REGION")

Re-scale t :

$$\tau = \epsilon^2 t \quad (0.30)$$

(Linearized, outer solution breaks down when $\sqrt{t} = o\left(\frac{1}{\epsilon}\right)$ or $t = o\left(\frac{1}{\epsilon^2}\right)$)

Now require a shock thickness, on the inner scale, to be order unity:

$$\xi = \epsilon(x - \sqrt{\gamma t}) = \epsilon\sqrt{2\beta X} \quad (0.31)$$

Expansion

$$\mu^i = \mu_0^i + \epsilon\mu_1^i + \epsilon^2\mu_2^i + \dots \quad (0.32)$$

Substitute into Navier-Stokes equations, and obtain Burgers' equation:

$$\mu_\tau^i + \frac{1}{2}(\gamma + 1)\mu^i \mu_\xi^i = \frac{1}{2}\beta\mu_{\xi\xi}^i \quad (0.33)$$

Boundary conditions (matching principle):

$$(\mu^i)^o = (\mu^o)^i \quad (0.34)$$

Initial conditions

$$\mu^i(\xi, 0) = 0, \xi > 0 \quad (0.35)$$

$$\mu^i(\xi, 0) = 1, \xi < 0 \quad (0.36)$$

Thus on the inner scale (large times) we have an initial value problem.

Consider the transformation

$$\mu^i = -\frac{2\beta}{(\gamma + 1)} \frac{\psi_\xi}{\psi} \quad (0.37)$$

Burgers' equation becomes:

$$\psi_\tau = \frac{1}{2}\beta\psi_{\xi\xi} \quad (0.38)$$

(Heat conduction equation)

$$\psi(\xi, 0) = \exp\left(-\frac{(\gamma + 1)}{2\beta}\xi\right), \xi < 0 \quad (0.39)$$

$$\psi(\xi, 0) = 1, \xi > 0 \quad (0.40)$$

Match inner and outer solutions using the asymptotic matching principle (not the limit matching principle).

Composite solution, μ^c

$$\mu^c = \mu^i + \mu^o - (\mu^i)^o \quad (0.41)$$

$$\mu^c = \mu^i \mu^o / (\mu^o)^i \quad (0.42)$$

MIT OpenCourseWare
<https://ocw.mit.edu/>

16.121 Analytical Subsonic Aerodynamics
Fall 2017

For information about citing these materials or our Terms of Use, visit: <https://ocw.mit.edu/terms>.